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### SECTION 7. Mechanics and machine construction.

## CALCULATION OF VON MISES STRESS AT PLASTIC DEFORMATION OF A STEEL BUSHING

**Abstract:** Stress condition of a steel bushing after short-term plastic deformation is presented in the article. The analytical formula for determining of von Mises stress at plastic deformation of the steel bushing is obtained.

**Key words:** stress, a bushing, tensor, a model, plastic deformation.

**Language:** English

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### Introduction

Hollow metal parts (for example, bushings and liners) are exposed to plastic deformation under external loads during operation [1]. Stresses of various kinds occur in material of the part. A calculation of a value and a distribution in a volume of stresses is carried out taking into account the physical and mechanical properties of material, sizes of the part, action time of loads and other conditions of plastic deformation. A number of multidirectional stresses in material of the deformed part can be presented by von Mises stress (equivalent stress) [2]. The calculated value of von Mises stress must be less than maximum allowable stress in material of the part in comparing. Safety margin of material at short-term plastic deformation of the thin-walled part has determined from the ratio of two stresses. Let us consider stress-strain condition of the thin-walled part in condition of constant radial external force application.

### Materials and methods

The calculation of von Mises stress was performed in the COMSOL Multiphysics (Structural Mechanics Module) software environment [3].

The three-dimensional solid-state bushing model was exposed to plastic deformation. Quality structural steel 1045 (UNS G10450) was accepted as material of the bushing model [4]. The outer diameter of the bushing model was accepted by the value of 40 mm, the inner diameter of the bushing model was accepted by the value of 30 mm. Constant distributed force of 1 kN by duration of 1 s acted on the outer cylindrical surface of the steel bushing model. Load was absent in axial direction. The steel bushing model was fixed (the inner cylindrical surface).

The following conditions were accepted for the calculation of stress condition of the steel bushing model at plastic deformation (1 – 11):

1. Solid mechanics

$$\rho \frac{\partial^2 u}{\partial t^2} = \nabla \cdot FS + Fv \quad (1)$$

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$$F = I + \nabla u \quad (2)$$

where  $\rho$  is density;  $u$  is displacement field;  $t$  is time;  $\nabla$  is gradient;  $F$  is force;  $S$  is the second Piola-Kirchhoff stress tensor [5];  $F_V$  is load defined as force per the unit volume;  $I$  is unit tensor.

### 2. Linear elastic material

$$\rho \frac{\partial^2 u}{\partial t^2} = \nabla \cdot S + F_V \quad (3)$$

$$S = S_{ad} + C : \varepsilon_{el} \quad (4)$$

$$\varepsilon_{el} = \varepsilon - \varepsilon_{inel} \quad (5)$$

$$S_{ad} = S_0 + S_{ext} + S_q \quad (6)$$

$$\varepsilon_{inel} = \varepsilon_o + \varepsilon_{th} + \varepsilon_{hs} + \varepsilon_{pl} + \varepsilon_{cr} \quad (7)$$

$$\varepsilon = \frac{1}{2} [(\nabla u)^T + \nabla u] \quad (8)$$

where  $S_{ad}$  is additive stress [6];  $C$  is the fourth-order elasticity tensor [7];  $\varepsilon_{el}$  is elastic strain;  $\varepsilon$  is total strain tensor;  $\varepsilon_{inel}$  is inelastic strain;  $S_0$  is initial stress;  $S_{ext}$  is external stress;  $S_q$  is stress (viscous damping);  $\varepsilon_o$  is initial strain;  $\varepsilon_{th}$  is thermal strain;  $\varepsilon_{hs}$  is hygroscopic strain;  $\varepsilon_{pl}$  is plastic strain;  $\varepsilon_{cr}$  is creep strain;  $T$  is temperature.

### 3. Boundary load

$$S \cdot n = F_A \quad (9)$$

$$F_A = \frac{F_{tot}}{A} \quad (10)$$

where  $n$  is outward unit normal vector;  $F_A$  is load defined as force per the unit area;  $F_{tot}$  is total force;  $A$  is the cross section area.

### 4. Fixed constraint

$$u = 0 \quad (11)$$

Mesh statistics: minimum element quality – 0.2487, average element quality – 0.7027, tetrahedral elements – 3227, triangular elements – 1304, edge elements – 132, vertex elements – 8.

## Results and discussion

Calculated von Mises stress in the steel bushing model after plastic deformation is presented in the Fig. 1.

Maximum equivalent stress of material was distributed in surface layers of a hole after removing of external load from the outer surface of the bushing model. Maximum von Mises stress, when considering of the end surface of the bushing model, was found in the second and the fourth quarters of the XY coordinate plane. The volume of the bushing model, which was located in the first quarter of the XY coordinate plane, was less exposed to equivalent stress. Von Mises stress reached the value up to 950 kN/m<sup>2</sup>.

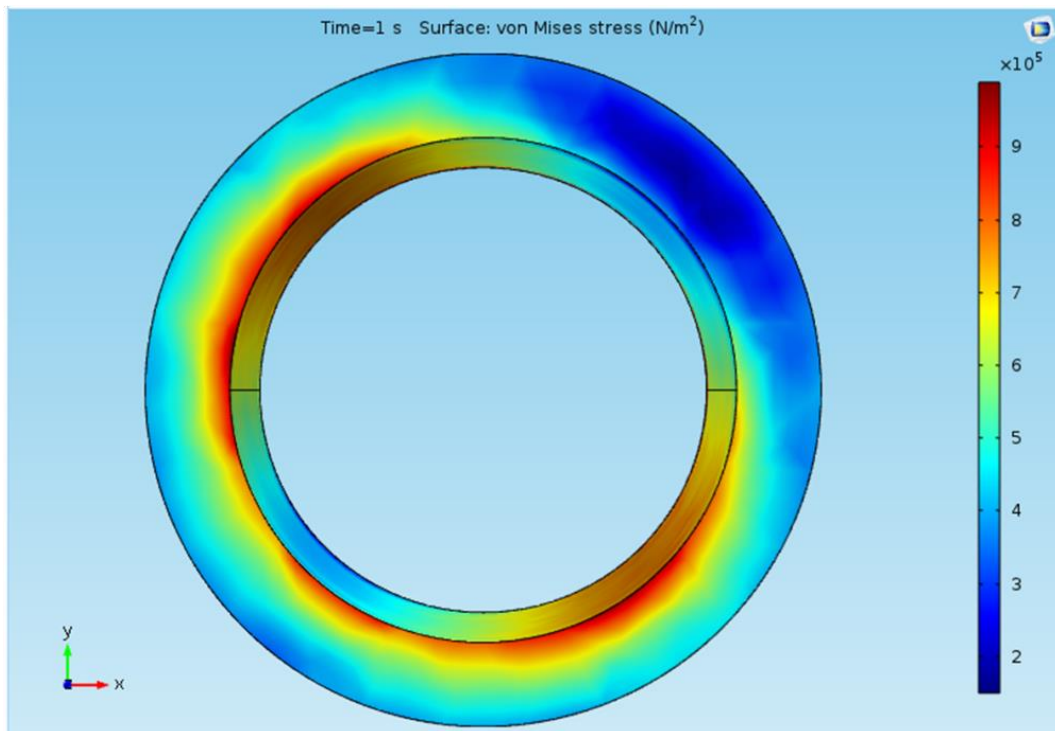


Figure 1 – Von Mises stress in the steel bushing model after plastic deformation.

Maximum allowable stress (tensile strength) of quality steel 1045 is 600000 kN/m<sup>2</sup>. This means that material of the bushing model can withstand external loads by the value in 600 times more than external

load applied in accordance with conditions of the problem.

Von Mises stress can be presented in the mathematical form (12)

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$$\begin{aligned}
 \sigma_{vonMises} = & \left[ \left( \frac{1.5 \left( \frac{E}{(1+\nu)(1-2\nu)} \right) \left( (1-\nu) \cdot \nabla u + \nu \cdot \nabla v + \nu \cdot \nabla w \right) - \frac{E}{(1+\nu)(1-2\nu)} \left( (1-\nu) \cdot \nabla u + \nu \cdot \nabla v + \nu \cdot \nabla w \right) + (\nu \cdot \nabla u + (1-\nu) \cdot \nabla v + \nu \cdot \nabla w) + (\nu \cdot \nabla u + \nu \cdot \nabla v + (1-\nu) \cdot \nabla w)}{3} \right)^2 \right. \\
 & \left. + 2 \left[ \frac{E}{1+\nu} (\nabla v + \nabla u) \right]^2 + 2 \left[ \frac{E}{1+\nu} (\nabla w + \nabla u) \right]^2 \right. \\
 & + \left( \frac{\frac{E}{(1+\nu)(1-2\nu)} (\nu \cdot \nabla u + (1-\nu) \cdot \nabla v + \nu \cdot \nabla w) - \frac{E}{(1+\nu)(1-2\nu)} \left( (1-\nu) \cdot \nabla u + \nu \cdot \nabla v + \nu \cdot \nabla w \right) + (\nu \cdot \nabla u + (1-\nu) \cdot \nabla v + \nu \cdot \nabla w) + (\nu \cdot \nabla u + \nu \cdot \nabla v + (1-\nu) \cdot \nabla w)}{3} \right)^2 \\
 & \left. + 2 \left[ \frac{E}{1+\nu} (\nabla w + \nabla v) \right]^2 \right)^{0.5} \quad (12)
 \end{aligned}$$

where  $E$  is the Young's modulus;  $\nu$  is the Poisson's ratio;  $\nabla u, \nabla v, \nabla w$  are displacement gradients.

planes, since the coefficients  $\left[ \frac{E}{1+\nu} (\nabla w + \nabla u) \right]^2$  and

$\left[ \frac{E}{1+\nu} (\nabla w + \nabla v) \right]^2$  in the formula (12) are equal.

### Conclusion

Evenly distributed radial force leads to asymmetric stresses in material of the bushing model. Stresses symmetry in material of the bushing model is observed in the  $XZ$  and  $YZ$  coordinate

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